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Progress report on: the Peltier-cooled frostpoint hygrometer (PCFH); and QCLAS and FLASH-B

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Peltier Cooled Frostpoint Hygrometer (PCFH): Report on 2018 Activities & Status

slides adapted from

GAW-CH Landesausschuss Spring Meeting, 14 May 2019, Empa Dübendorf, Switzerland

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PCFH Fundamentals

- Peltier-cooled mirror hygrometer instead of cryogenically-cooled like FPH or CFH (phase down
 agreement of the Kigali Amendment to the Montreal Protocol, 2016).
- Research and test instrument: includes two independently cooled mirrors and boards.





Field Campaigns at Lindenberg

- 1st Field Campaign in Lindenberg, 23 .. 27 July 2018: **2** flights
- 2nd Field Campaign in Lindenberg, 10...14 December 2018: **3** flights

Aims of the Field Campaigns

- No frost point measurements!
- Approval of basic concepts (optical, thermal, electrical, telemetry)
- Characterization of Thermal Electric Cooler (TEC) :
 - in lab
 - in flight environment
- Characterization of condensate formation and detection (mirror reflectivity monitoring)
- Gather data for instrument numerical model implementation

1st Field Campaign: Flights & Payloads

PCFH Flight 1

2018-0725-02UT_LI_LI236 (regular science)

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- 2018-0725-02UT_LI_PCFH001
- Two separate RS41 telemetries



- Power supply: sufficient for one flight duration
- Robust: only battery replacement required after recovery. Contemport
- Telemetry issues

PCFH Flight 2

 2018-0725-20UT_LI_PCFH002: ECC O3, COBALD, CFH & PCFH







PCFH



1st Field Campaign: Selected Results TEC Performance





Frost point temperature reached only up to tropopause level.

 Heat load was underestimated → need lighter heat sink, protection (radiation shield) and better thermal coupling.



PCFH Setup Revisions



1st Flights' Version: not enough heat dissipation



- More space and better thermal coupling
 → better heat dissipation
- new electronics design
 → higher TEC Current achieved

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2nd Field Campaign: Flights & Payloads







PCFH Flight 3 & 4 ("simultaneous" launch)

- 2018-1212-16UT_LI_LI246-CFH two single stage TECs
- 2018-1212-16UT_LI_LI246-COB single & double stage TECs

All Flights

telemetry problem resolved

PCFH Flight 5

 2018-1213-22UT_LI_LI247 single & double stage TECs ETTH Eldgenössische Technische Hachschule Zürich Swiss Federal Institute of Technology Zurich

2nd Field Campaign Results: Cooling performance during ascent



 $T_{hot} \cong T_{air} + 10 \text{ K}$

Summary and Project Schedule

- Double stage TEC with increased spacing and with coupling with cupper block provides adequate temperature range
- Next steps:
 - Improve Peltier hot side to heat sink coupling to minimize T_{hot} T_{air} (Nov-Dec 2019)
 → e.g. use larger cupper block
 - Numerical instrument model (2019- early 2020)
 - Analyze ice formation dynamics and its effects on mirror reflectivity to introduce timing in the numerical instrument model
 - Add and test controller scheme in the numerical model
- Field verification of the design changes
 - Two campaigns in Lindenberg during November / December 2019
 - Four dual payload launches from Ny-Alesund in February 2020
- Test Flights within Monitoring Networks is postponed by approximately one year (as of May 2019).

Quantum Cascade Laser based Absorption Spectroscopy (QCLAS) Balloon Instrument

alides from Manuel Graf, EMPA, Switzerland



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Quantum Cascade Laser based Absorption Spectroscopy (QCLAS)

Principles

- Direct absorption spectroscopy
- Excitation of a rovibrational transition at 1662 cm⁻¹ (6.01 μ m)
- Open path operation (L=6 meters)
- Novel circular multipass cell allows simple & robust setup¹⁾

$$P_r = C * P_0 * \exp\left(-\left(N_{H_2O} * \sigma_{abs,H_2O} + \epsilon\right) * L\right) + P_b$$

Method





The Instrument

Specs

size	30 x 20 x 10 cm ³
mass	3 kg
gas exchange (5 m/s ascent speed)	40 l/s
power consumption	15 W
measurement precision	<1%
measurement accuracy	tba



- Fully integrated electronics and batteries
- Thermal stabilization with phase change materials (PCM)
- 2 h autonomous operation



bottom view

Alexey Lykov, Sergey Khaykin, Central Aerological Observatory, Russia CNRS/INSU, LATMOS-IPSL, France

Fluorescence Lyman-Alpha Stratospheric Hygrometer for Balloon (FLASH-B)



slides from Sergey Khaykin



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www.flash-b.ru Method of measurement and optical layout

The fluorescent method is based on the photodissociation of H2O molecules under UV lamp light at wavelengths below 137 nm and subsequent fluorescent relaxation of the exited OH* radical produced.





Optical layout of the fluorescent hygrometer (FLASH) for balloon

The instrument uses the open layout, where the optics is looking directly into the outside air. Therefore, measuring only <u>at night</u>. <u>For stratosphere, descent data only</u>

Specifications



Flights in Dolgoprudny

Altitude (km)



Tropospheric comparison with RS41 and M-10



RH (%)

3 consecutive flights of the same FLASH sonde in Dolgoprudny with a series of calibrations before and after each flight

- Consistent water vapour vertical profiles in the stratosphere and good agreement with M10 and RS41 in the free troposphere
- Small change of sensitivity factor (<10 %) in between the calibrations/flights

Application on different platforms



Thank you for your attention! Questions?

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Backup slides





Flight preparation

Plug-and-play concept





Sealant recommended against contamination in stratosphere

- 1. Recharge battery;
- 2. Fixing face down;
- 3. Clean optics;
- 4. Connect to radiosonde;
- 5. Switch on;
- 6. Check the data incoming;
- 7. Open cap;
- 8. Ready to fly!







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Calibration facility







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PCFH 2018 Activities & Status, GAW-CH Lar Gentral Action Bieser Vatory, ROSHYDROMET

Calibration data

The calibration fit function is linear in the pressure range of 30 - 150 hPa and water vapor mixing range of 1 - 500 ppmv.





The total relative error of the calibration amounts to 4%.



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Calculation formula





FLASH-B error budget

Calibration error

- MBW373L dew point uncertainty (0.1 K)
- Pressure error (conversion to mixing ratio)
- Non-linearity of calibration curve
- Outgassing within calibration chamber
- Random error (operator-related factors)

Total relative error of calibration is estimated at 4%

Measurement error

- Instability of Lyman-alpha emission, including temperature-related drifts (<3%)
- Random error (5.5% precision for 4 s integration)
- Detection limit 0.1 ppmv

Total uncertainty (calibration error + 1σ precision) is below 10% at stratospheric conditions.

(1σ) Comparison against other hygrometers and in-flight repeatability fully confirms the estimated

